APPROVAL CERTIFICATE

EASA.21J.030

Pursuant to Regulations (EU) 2018/1139 and (EU) 748/2012 and subject to the conditions specified below, the Agency hereby certifies

Grob Aircraft SE

Lettenbachstrasse 9
D-86874 Tussenhausen
Germany

as a DESIGN ORGANISATION

approved according to Part 21, Section A, Subpart J.

CONDITIONS :

1. The approval is limited to that specified in the enclosed Terms of Approval, and
2. This approval requires compliance with the procedures specified in the Design Organisation Handbook, reference Entwicklungsbetriebshandbuch, in the latest revision, and
3. This approval is valid whilst the approved Design Organisation remains in compliance with Part 21, Section A, Subpart J.
4. Subject to compliance with the foregoing conditions, this approval shall remain valid until surrendered or revoked.

For the **European Union Aviation Safety Agency,**
Date of issue: 10 July 2025

Carmela BOSSO
Section Manager - Design Organisations & Policy Issues

Terms of Approval

Design Organisation Approval Certificate

EASA.21J.030

1 Scope

This Design Organisation Approval is applicable for the scope defined in Annex A for design work with regard to the airworthiness, operational suitability and environmental characteristics of the products.

2 Privileges

a) (Reserved)

b) (Reserved)

c) The holder of this design organisation approval shall be entitled, within the scope of this terms of approval, and under the relevant procedures of the design assurance system:

1. to classify changes to a type-certificate or to a supplemental type-certificate and repair designs as “major” or “minor”;
2. to approve minor changes to a type-certificate or to a supplemental type-certificate and minor repair designs;
3. [Not applicable];
4. [Not applicable];
5. to approve certain major repair designs under Part 21, Section A, Subpart M to products or auxiliary power units (APUs);
6. to approve for certain aircraft the flight conditions under which a permit to fly can be issued in accordance with point 21.A.710(a)(2), except for permits to fly to be issued for the purpose of point 21.A.701(a)(15);
7. to issue a permit to fly in accordance with point 21.A.711(b) for an aircraft it has designed or modified, or for which it has approved, in accordance with point 21.A.263(c)(6), the flight conditions under which the permit to fly can be issued, and where the holder of this design organisation approval itself:
   (i) controls the configuration of the aircraft, and
   (ii) attests conformity with the design conditions approved for the flight;
8. [Not applicable];
9. [Not applicable].

3 Obligations

The holder of this design organisation approval shall, within the scope of this terms of approval:

a) maintain the handbook required under point 21.A.243 in conformity with the design assurance system;
b) ensure that this handbook or the relevant procedures included by cross-reference are used as a basic working document within the organisation;
c) determine that the design of products, or changes or repairs thereto comply with the applicable specifications and requirements and have no unsafe features;
d) provide the Agency with statements and associated documentation confirming compliance with point (c), except for approval processes carried out in accordance with point 21.A.263(c);
e) provide to the Agency data and information related to the actions required under point 21.A.3B;
f) determine, in accordance with point 21.A.263(c)(6), the flight conditions under which a permit to fly can be issued;
g) establish, in accordance with point 21.A.263(c)(7), compliance with points (b) and (e) of point 21.A.711 before issuing a permit to fly to an aircraft;
h) designate data and information issued under the authority of the approved design organisation within the scope of its terms of approval as established by the Agency with the following statement: “The technical content of this document is approved under the authority of the DOA ref. EASA. 21J.030”.
i) comply with Part 21, Section A, Subpart A.

Date of issue: 10/07/2025

Carmela BOSSO
Section Manager - Design Organisations & Policy Issues

Annex A

Scope of work

			TC	STC	major changes	minor changes	major repairs	minor repairs	flight conditions	permit to fly
(Powered) sailplane										
	All scope (TCH)									
		All areas	■		■	■	■	■	■	■
	Avionics (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Cabin (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Electrical Systems (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Flight (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Hydro-Mechanical Systems (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Powerplant and Fuel Systems (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Structures (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
Piston engine										
	All scope (TCH)									
		All areas	■		■	■	■	■		
Small aeroplane										
	All scope (TCH)									
		All areas	■		■	■	■	■	■	■
	Avionics (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Cabin (non-TCH activity)									
		All areas		■	■	■	■	■	■	■
	Electrical Systems (non-TCH activity)									

		All areas	■ ■ ■ ■ ■ ■ ■
	Environmental Control Systems (non-TCH activity)		
		All areas	■ ■ ■ ■ ■ ■ ■
	Flight (non-TCH activity)		
		All areas	■ ■ ■ ■ ■ ■ ■
	Hydro-Mechanical Systems (non-TCH activity)		
		All areas	■ ■ ■ ■ ■ ■ ■
	Powerplant and Fuel Systems (non-TCH activity)		
		All areas	■ ■ ■ ■ ■ ■ ■
	Structures (non-TCH activity)		
		All areas	■ ■ ■ ■ ■ ■ ■
Turbine engine			
	Powerplant and Fuel Systems (non-TCH activity)		
		Fuel systems	■ ■ ■

Legend:

	Title for category of product
	Title for design scope
	Title for design area

■	Within scope
	Outside scope

List of products

Product	Design Activity	Types
(Powered) sailplane	TC	TCDS ref. DE TC 817: G 109
Piston engine	TC	TCDS ref. DE 4601: GROB 2500
Small aeroplane	TC	TCDS ref. EASA.A.364: G 115 TCDS ref. EASA.A.075: G 120 TCDS ref. EASA.A.565: G 120TP TCDS ref. DE 2066: G 520

Limitations

	Limitations common to all products and activities
	Operational Suitability Data excludes the OSD constituents Flight Crew Data, Cabin Crew Data, Simulator Data and Maintenance Certifying Staff Data The privilege under paragraph 2(c)(5) is limited to the approval of the design of major repairs to products for which the DOA holds the type-certificate or the supplemental type-certificate.

Product	Limitations particular to each product
(Powered) sailplane	For non-TCH activity:
	For TCH activity:
Piston engine	For TCH activity:
Small aeroplane	For non-TCH activity:
	For TCH activity:
Turbine engine	For non-TCH activity: Changes limited to specific systems as defined in DO Handbook